

Code: ME4T3

II B.Tech - II Semester – Regular Examinations – May 2016

**IC ENGINES AND GAS TURBINES
(MECHANICAL ENGINEERING)**

Duration: 3 hours

Max. Marks: 70

PART – A

Answer *all* the questions. All questions carry equal marks

11x 2 = 22 M

1. a) Define the time loss factor.
- b) What is the cubic capacity of the engine?
- c) Define ‘Scavenging’.
- d) What is the transposition rate in the flame front propagation?
- e) What is pre-ignition?
- f) What are the primary reference fuels to rate the CI engine fuels?
- g) List at least four methods to find frictional power.
- h) Define Indicated mean effective pressure.
- i) What are the desired properties of any alternate fuel for SI engines?
- j) What is open cycle gas turbine power plant ?
- k) Draw the T-S diagram for Jet engine.

PART – B

Answer any **THREE** questions. All questions carry equal marks. 3 x 16 = 48 M

2. a) What is the difference between air standard cycle approximation, fuel-air cycle approximation and real cycle performance? 8 M
- b) What type engine is used for the two wheeler (Bajaj Pulsur)? Explain its working in detail. 8 M
3. a) Describe the different phases of combustion phenomenon in SI engines with suitable sketches. 12 M
- b) CI engines will accept the lean mixtures. Justify the statement. 4 M
4. a) What are the desirable properties for the SI engine fuel? If an alternative fuel with high octane number is used, what will be its effect on combustion? 5 M
- b) Find the engine dimensions of a 2-cylinder, 2-stroke I.C. engine from the following data:
Engine speed= 4000 rpm,
Volume efficiency = 0.77,
Mechanical efficiency = 0.75,
Fuel consumption = 10 Lit/hr $S = 0.73$,
Air fuel ratio = 18:1,

Piston speed = 100 m/s,

Indicated mean effective pressure = 5bar.

Find also brake power. Take R of gas mixture as

281 J/Kg k at S.T.P.

11 M

5. A gas turbine unit has a pressure ratio of 6:1 and maximum cycle temperature of 610⁰C. The isentropic efficiencies of the compressor and turbine are 0.80 and 0.82 respectively. Air enters the compressor at 15⁰C at the rate of 16kg/s.

Calculate

i) work required for compression

ii) power developed by the turbine

iii) Net power output in kilowatts

Take $C_p=1.005$ kJ/kgk and $\gamma=1.4$ for the compression process, and take $C_p=1.11$ kJ/kgk and $\gamma=1.333$ for the expansion process.

16 M

6. a) Describe the working principle of the turbo jet engine.

8 M

- b) What are the different thrust augmentation methods?

Explain.

8 M